

Development of a New Generation of High-Temperature Thermoelectric Unicouples for Space Applications

Direct Thermal-to-Electrical Energy Conversion
San Diego, CA
August 2006

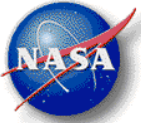
presented by

T. Caillat

co-authors

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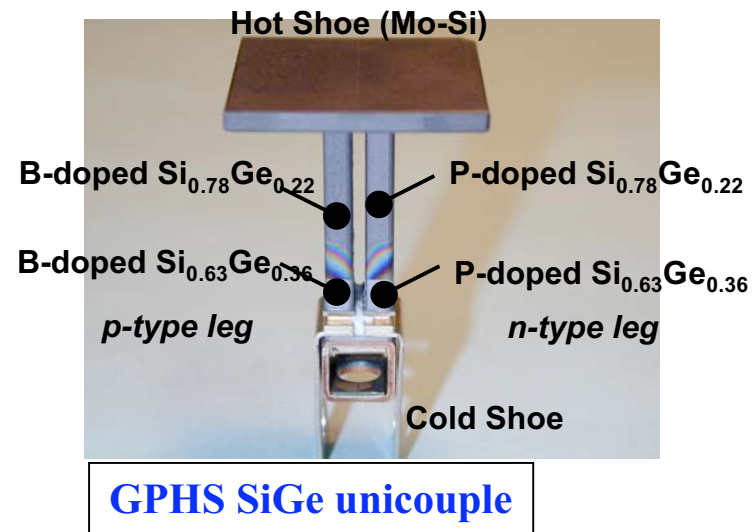
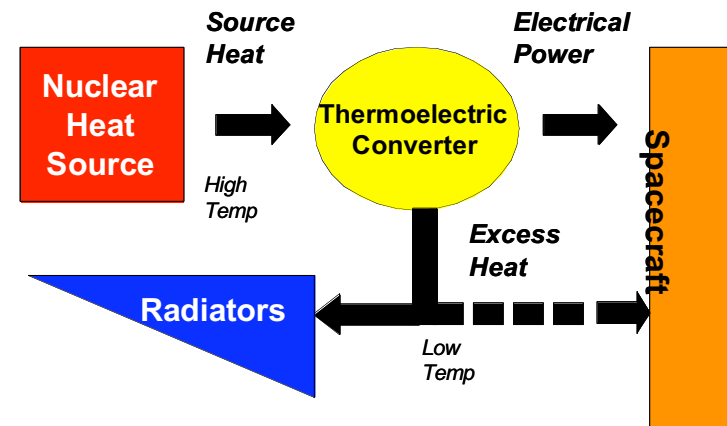
Jet Propulsion Laboratory/California Institute of Technology



What is a Radioisotope Thermoelectric Generator?

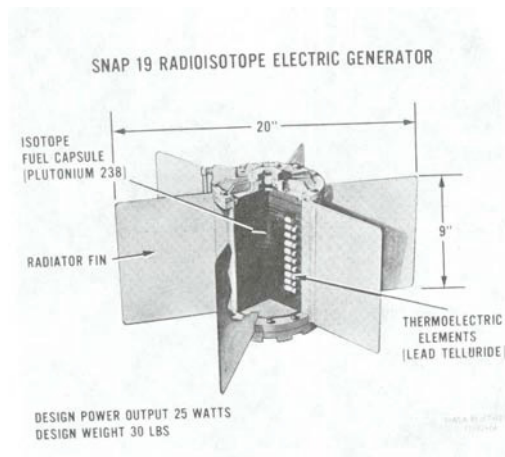
- Long life, high reliability DC electrical power source
- Converts heat from a radioisotope heat source to electricity
- Contains a nuclear radioisotope heat source, thermoelectric converter and a radiator
- Thermoelectric converter contains many couples connected in series and parallel networks
- Unicouples consist of two semiconductor thermoelectric legs (p and n) placed between hot and cold temperatures
- Efficiency of the couple depends on the thermoelectric properties of the uncouple (ZT) and temperature difference between hot and cold ends

$$\eta = \frac{T_H - T_C}{T_H} \frac{\sqrt{1 + ZT} - 1}{\sqrt{1 + ZT} + \frac{T_C}{T_H}} \quad Z = \frac{\alpha^2}{\rho\lambda}$$





Flight Demonstrated Radioisotope Thermoelectric Generators (3 Most Recently Flown Designs)



SNAP-19(PbTe RTG)
(1960-70's)

40.3 Watts (BOM)
6.2 % sytem efficiency
3 We/kg

22.86 cm (9.0 in) long
50.8 cm (20 in) dia
~13 kg (28.6 lb)
PbTe Thermoelectrics

Nimbus B-1/III, Pioneer 10/11,
Viking 1/2

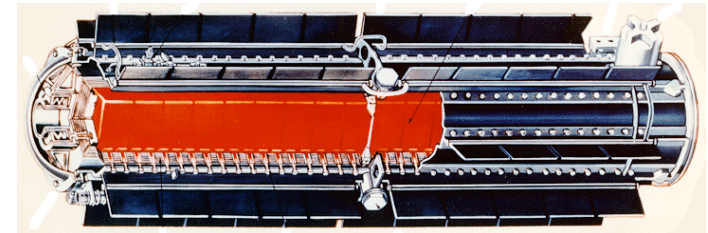


SiGe MHW RTG
(1970's)

158 We (BOM)
6.6 % system efficiency
4.2 We/kg

58.4 cm (23 in) long
39.7 cm (15.64 in) dia
38 kg (83.7lb)
SiGe Thermoelectrics

LES 8/9, Voyager 1/2



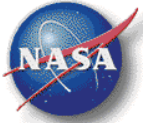
SiGe GPHS RTG
(1980-2006)

285 We (BOM)
6.8% system efficiency
5.1 We/kg

114 cm (44.9 in) long
42.7cm (16.8in) dia
56 kg (123 lb)
SiGe Thermoelectrics

Galileo, Ulysses, Cassini
& New Horizons

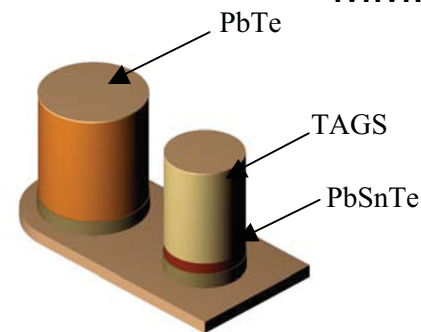
Pre-Decisional – For Discussion Purposes Only



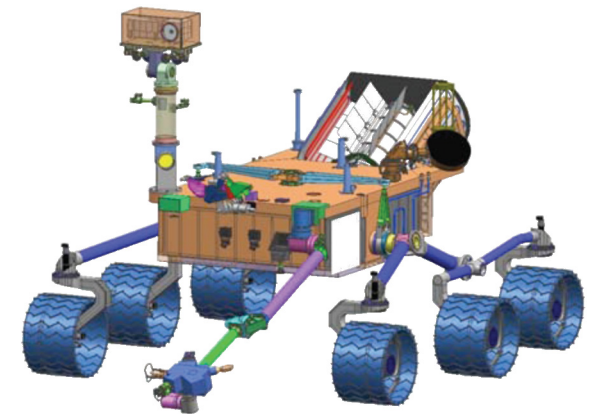
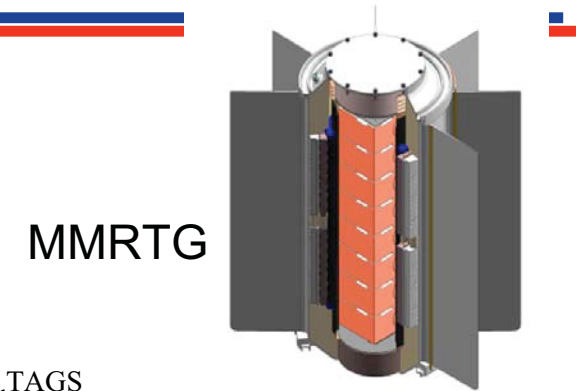
MMRTG Characteristics

- Electrical Power Output: ~ 123 W (BOL)
- Specific power: ~ 2.8 W_e/kg
- System Efficiency $\sim 6.2\%$
- Voltage 28 VDC
- In-space & surface operational capability
- Qualified for 0.2 g²/Hz random vibrations
- Mission life design ~ 14 years

- Mass: ~ 44.1 kg
- Dimensions (half that of GPHS-RTG):
 - ◆ Length ~ 64 cm
 - ◆ Diameter ~ 64 cm
- 8 GPHS modules
- Thermal Power Input ~ 2000 W (BOL)
- 768 PbSnTe/TAGS+ PbTe couples
 - ◆ $T_{\text{hot}} \sim 811$ K;
 - ◆ $T_{\text{cold}} \sim 483$ K

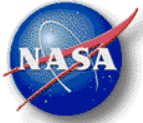


PbSnTe/TAGS+ PbTe couples



Mars Science Laboratory (MSL)

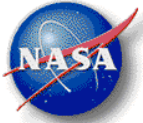
MMRTG Under Development For MSL And Other Future Deep Space And Surface Missions
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NASA Advanced RTG Needs

	Near Term (~2015)	Long Term (>2020)
Specific Power (W/kg)	6 - 8	> 10
Readiness	2015 - 2016	> 2020
Lifetime	> 14 years < 22% degradation	> 14 years < 22% degradation
Heat Source	Step 2 GPHS (8 to 12 units)	Step 2 GPHS (1 to 12 units)
System Efficiency (%)	10	13 - 15

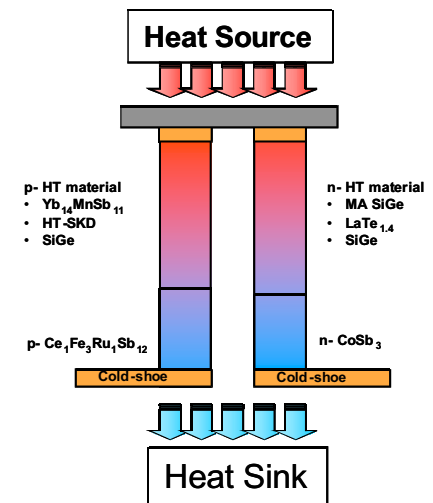
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■ Near Term Advanced RTG Development

Project

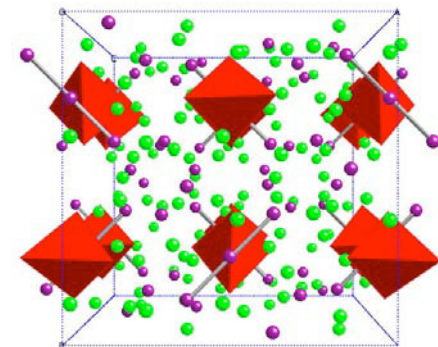
- ◆ Advanced Thermoelectric Converter (ATEC)
Development



■ Far Term Advanced RTG Development

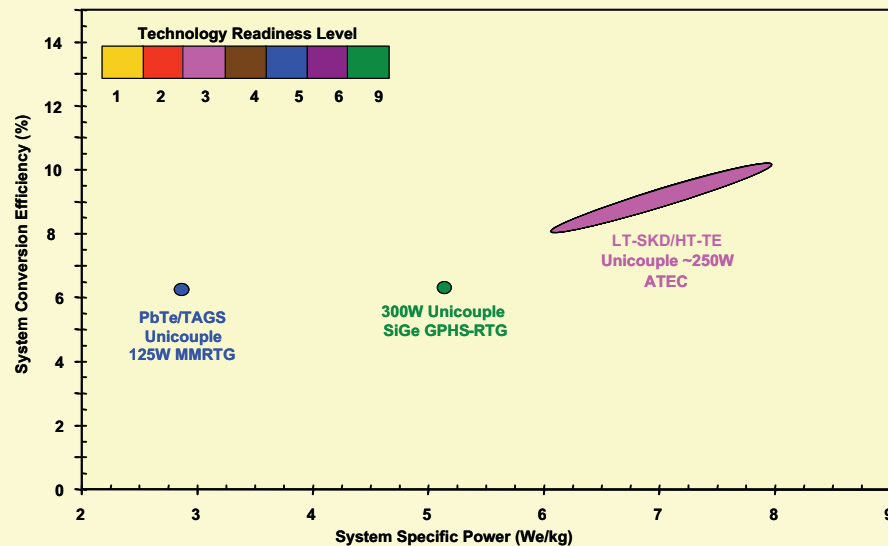
Tasks

- ◆ Si-Ge Nanocomposites
- ◆ Advanced Thermoelectrics R&T





Advanced Thermoelectric Converter (ATEC)



Major Objectives:

- Develop and demonstrate advanced thermoelectric converter capable of supporting a deep space RTG with 6-8 We/kg and 14 year lifetime.
- Develop and demonstrate couple with 11-12% conversion efficiency
- Demonstrate at least 1 year of lifetime operation on both couples and 4-couple modules
- Prediction of maximum of 22% power degradation (including isotope decay) over 14 years

Benefit: 110-180% RTG Specific Power Increase and 28 - 60% system conversion efficiency increase over MMRTG

PM: Rao Surampudi

PI: Thierry Caillat

Participating Organizations: JPL, GRC, USC, MSFC, Systems contractor

Milestones:

FY'06 Select primary & back-up high temperature TE materials

FY'07 Develop updated TE materials database

Validate couple power output within 10% of predict

FY'08 Validate 4-couple module power output within 10% of predict

Complete couple fabrication and assembly specification

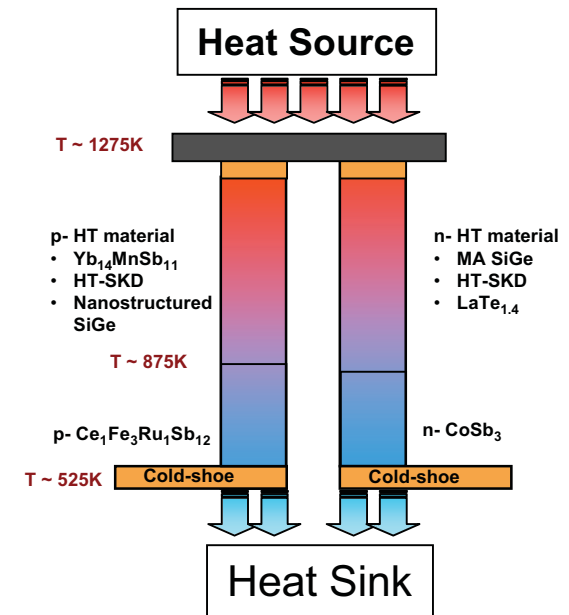
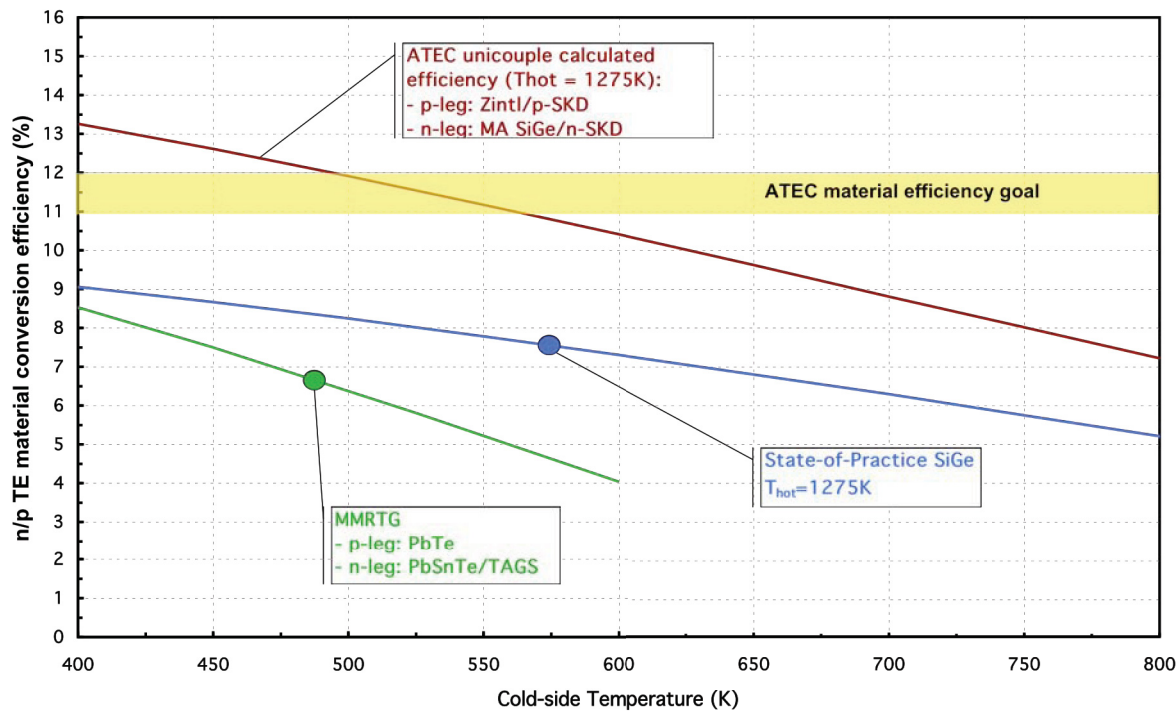
FY'09 Validate 1 year of couple life with < 0.33% degradation



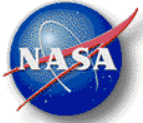
Projected Performance



- High ZT (>1.0) thermoelectric materials
 - ◆ Higher efficiency
- Segmented couples
 - ◆ Each segment optimized for maximum performance
- Large Delta T: (1275 to 525 K Operation)
 - ◆ Higher efficiency
- Sublimation Control
 - ◆ Aerogel
 - ◆ Metal/metal oxide coatings



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Synthesis Approach for Scalability



Spex High Energy Ball Mill

- Vial load capacity 10 – 15g
- Double vials system
- Back-and-forth shaking motion
- Balls impact with powder and vial
- ~ 1200rpm, vial swing ~ 5cm
- Making the force of the balls impact very high



Spex Mill



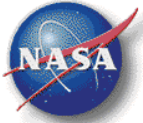
Planetary Low Energy Ball Mill

- Vial load capacity – few hundred grams
- Two or four vials system
- Planet-like movement of its vials
- Centrifugal force produced by the vials rotating around their own axes and produced by the rotating support disk both act on the vial content
- Low impact synthesis process



Planetary Mill

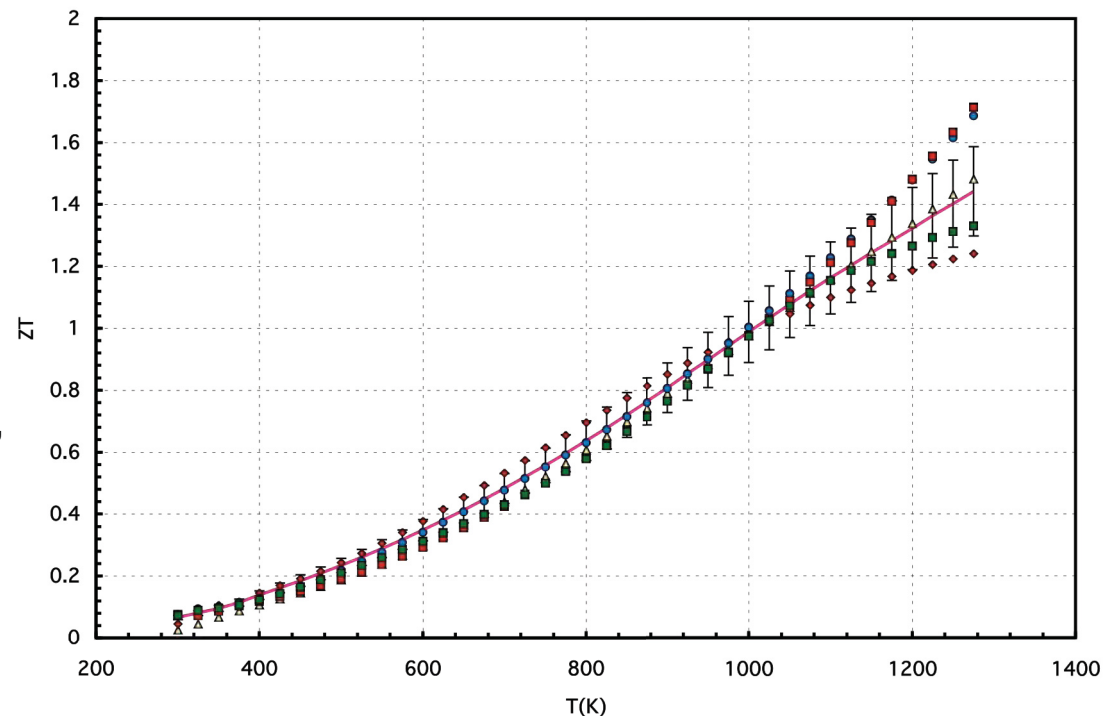




High Temperature Thermoelectric Materials Results: p- type Zintl



- Thermoelectric Properties
 - ◆ $ZT > 1$ at 1275K
- Synthesis process
 - ◆ Mechanical alloying
- Reproducibility
 - ◆ TE properties reproduced on >10 batches
- Scalability
 - ◆ 15g batches demonstrated
 - ◆ 50g batches by end of 2006
- Mechanical Properties
 - ◆ Initiated Young and shear modulus, Poisson's ratio, fracture toughness, flexural strength
- Sublimation
 - ◆ BOL ($\sim 5 \times 10^{-3}$ g/cm² /hr at 1275K)
- Segmentation
 - ◆ Co-hot-pressed several legs (promising)
- Some remaining challenges
 - ◆ Achieve sublimation goal of $\sim 10^{-7}$ g/cm² /hr through use of sublimation suppression coatings
 - ◆ Demonstrate TE property stability over time
 - ◆ Segmentation to low-T SKD material



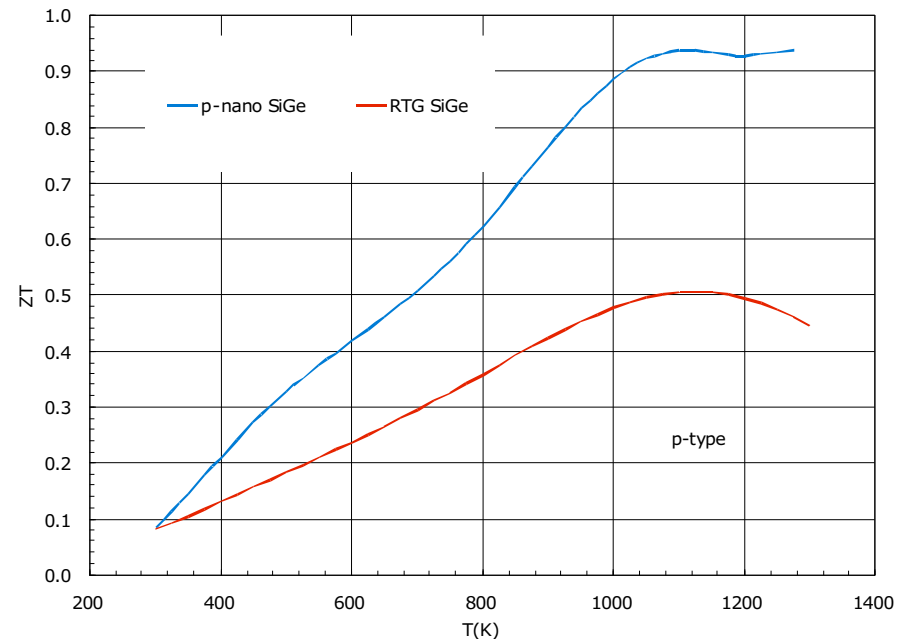
ZT values for Yb₁₄MnSb₁₁ material



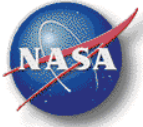
High Temperature Thermoelectric Materials: p - type NanoSiGe (MIT/JPL)



- Thermoelectric Properties
 - ◆ ZT approaching 1 at 1275K
- Synthesis process
 - ◆ Mechanical alloying
- Reproducibility
 - ◆ In progress
- High temperature Stability
 - ◆ Demonstrated stability of TE properties for 750 hrs at 1275K under MIT / JPL NRA task
- Scalability
 - ◆ 15g batches demonstrated
 - ◆ 50g batches by end of 2006
- Mechanical Properties
 - ◆ Will be initiated by end of 2006
- Sublimation
 - ◆ Plan to measure BOL sublimation and compare with RTG materials
 - ◆ Si_3N_4 coatings (GPHS-RTG SiGe)
- Segmentation
 - ◆ GPHS-RTG heritage – for metallization
 - ◆ Need to develop segmentation to low-T SKD if needed
- Remaining challenge:
 - ◆ Demonstrate batch to batch reproducibility and scalability
 - ◆ Further demonstrate TE property stability over time



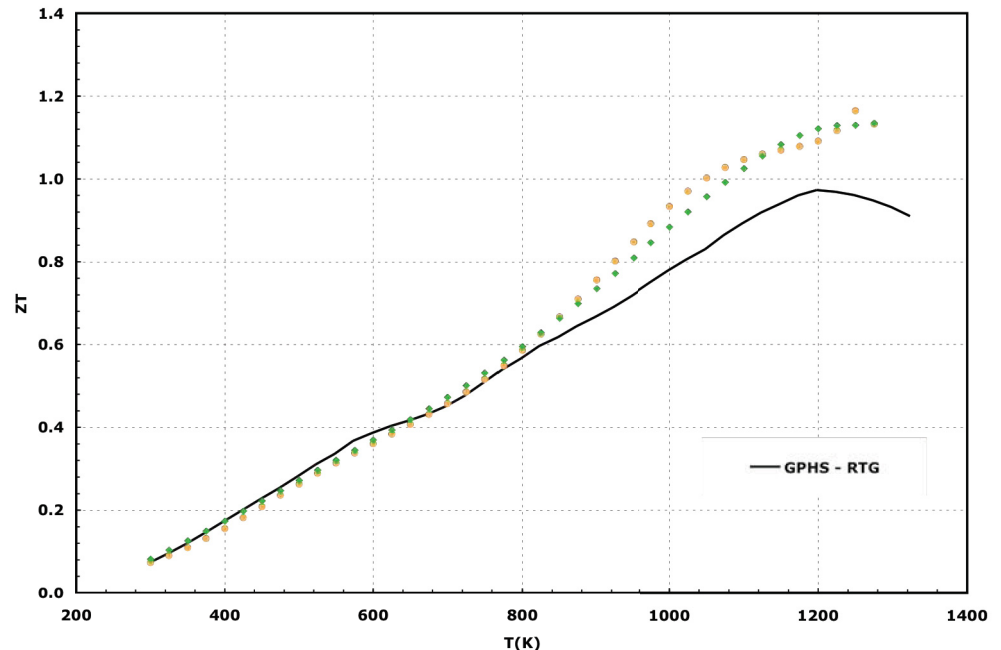
**Potential Backup
material for Zintl**



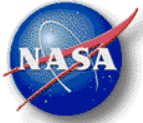
High Temperature Thermoelectric Materials: n - type SiGe



- Thermoelectric Properties
 - ◆ $ZT \sim 1.1$ at 1275K
- Synthesis process
 - ◆ Mechanical alloying
- Scalability
 - ◆ 15g batches demonstrated
 - ◆ 50g batches by end of 2006
- Reproducibility
 - ◆ $ZT \pm 10\%$ of 4 consecutive batches
- Sublimation
 - ◆ Si_3N_4 coatings used for GPHS-RTG SiGe
 - ◆ Plan to measure BOL sublimation and compare with GPHS-RTG SiGe
- Segmentation
 - ◆ RTG heritage – for metallization
 - ◆ Need to develop segmentation to low-T skutterudite
- Mechanical Properties
 - ◆ Measurements initiated
 - ◆ Expected to be similar to GPHS-RTG SiGe
- Issues / concerns
 - ◆ Achieve higher $ZT > 1$ at 1275K



Improved mechanically
alloyed SiGe



High Temperature Thermoelectric Materials

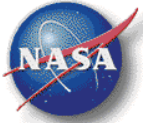
n - type $\text{LaTe}_{1.4}$



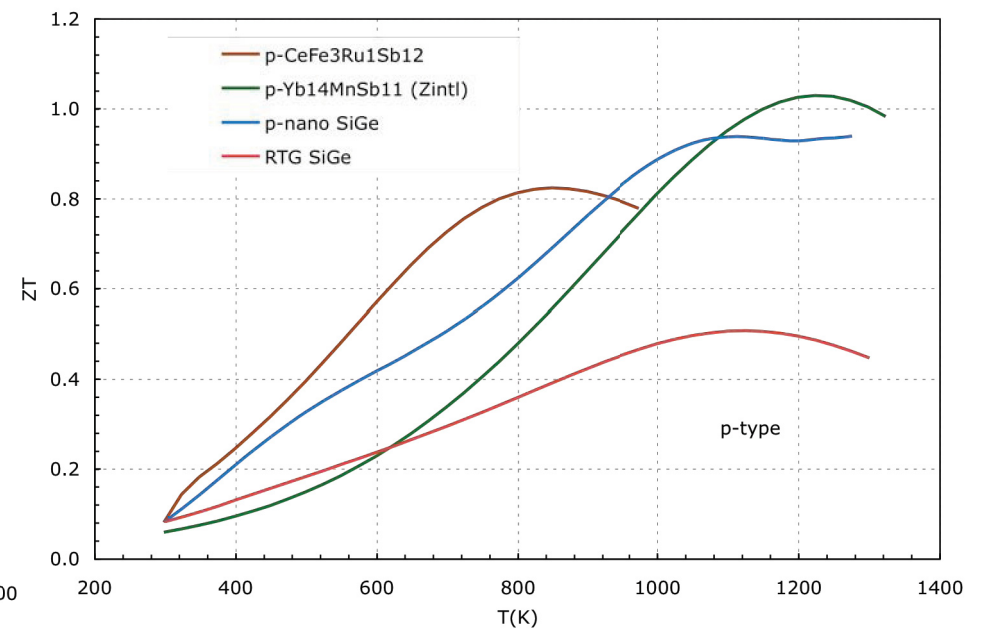
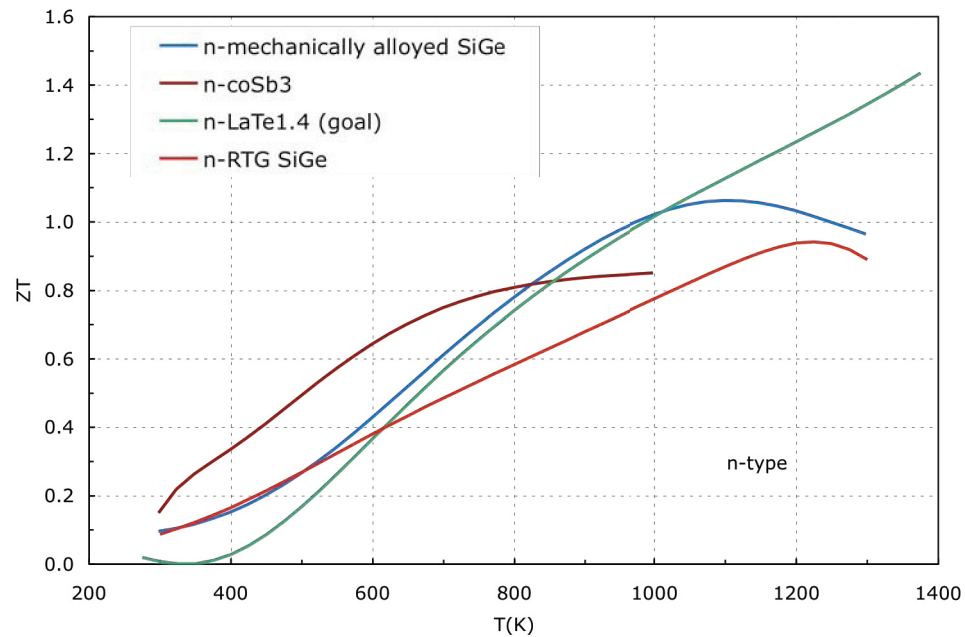
- Thermoelectric Properties
 - ◆ Not measured yet for optimized samples
 - ◆ Goal is $ZT \sim 1.2$ at 1275K
- Reproducibility
 - ◆ Not yet
- Scalability
 - ◆ 15g batches
- Mechanical Properties
 - ◆ Not yet
- Sublimation
 - ◆ Not yet
- Segmentation
 - ◆ Not yet
- Remaining challenges
 - ◆ Develop reproducible mechanical alloying synthesis process
 - ◆ Optimize TE properties by controlling La to Te ratio

N- $\text{LaTe}_{1.4}$

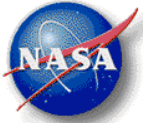
- Potential backup for mechanically alloyed n- SiGe
- Pay back is potentially high since this material offers high $ZT > 1$ at high temperatures



ZT summary



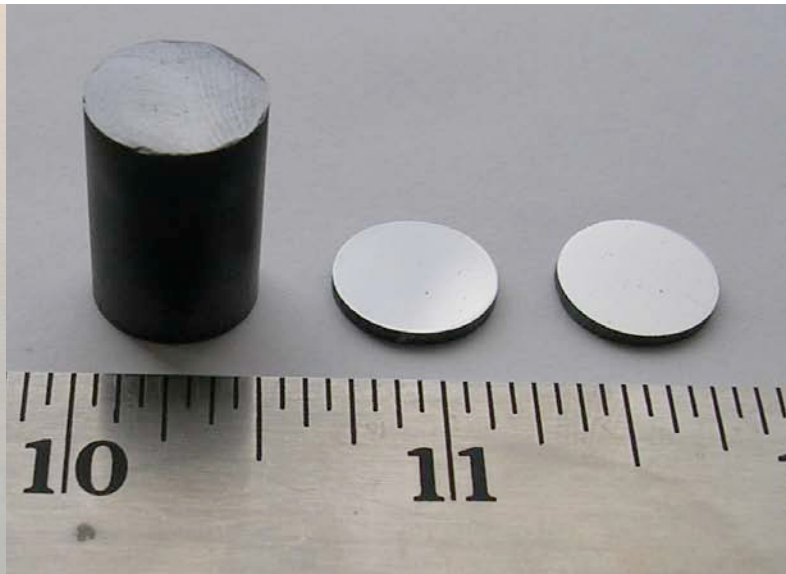
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Low-T skutterudite development



Mechanical Properties
40mm x 7mm puck

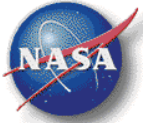


Sublimation Suppression Control Disks
12mm x 1.5mm disks

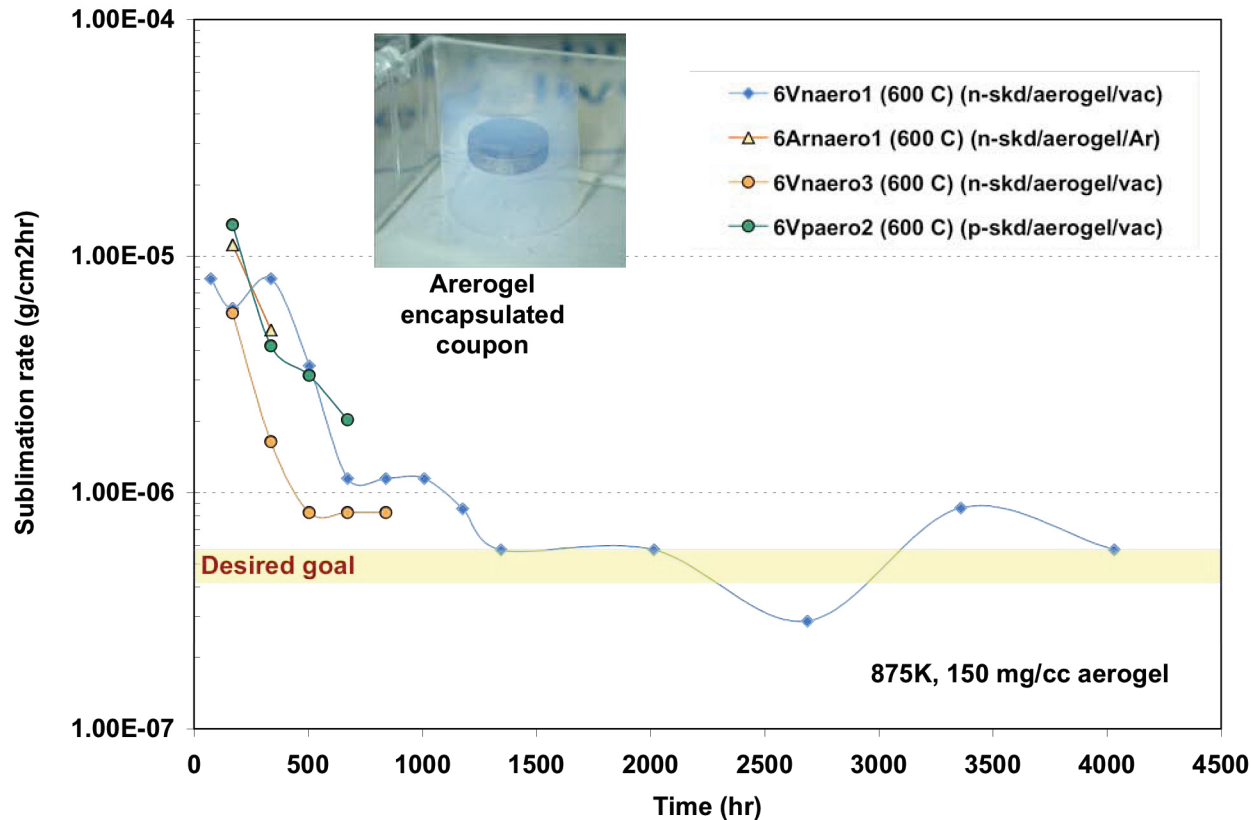


CTE and Compression
6.4mm x 25mm

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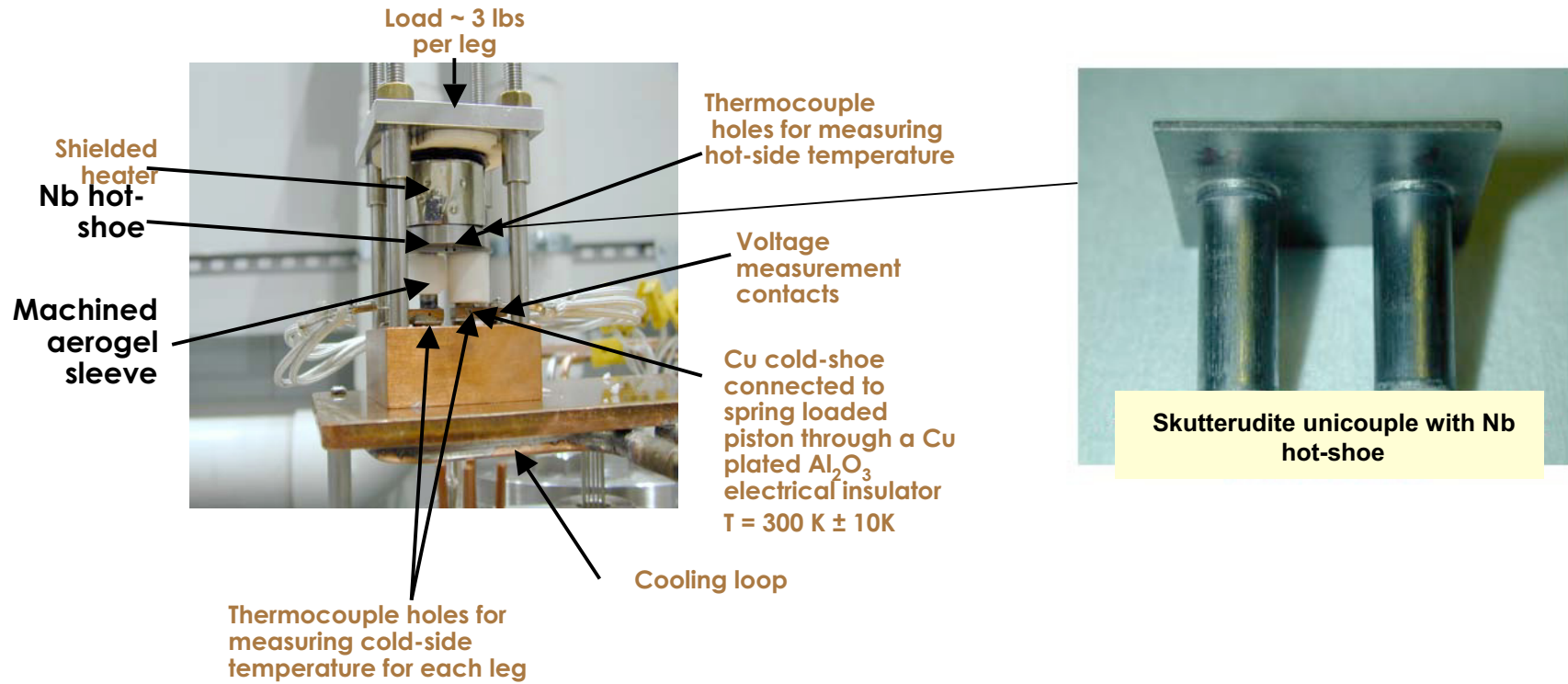


Sublimation Rates Life Tests at 875 K for n- and p-type low-T SKD Aerogel Encapsulated Coupons



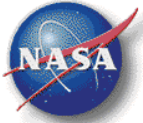
- Demonstrated that the desired sublimation rate ($< 5 \times 10^{-7}$ g/cm²hr) for 14 years of operation can be achieved up to 875K for aerogel-encapsulated low-T skutterudites after up to 4000 hours of testing

Low-T skutterudite -uncouple performance testing

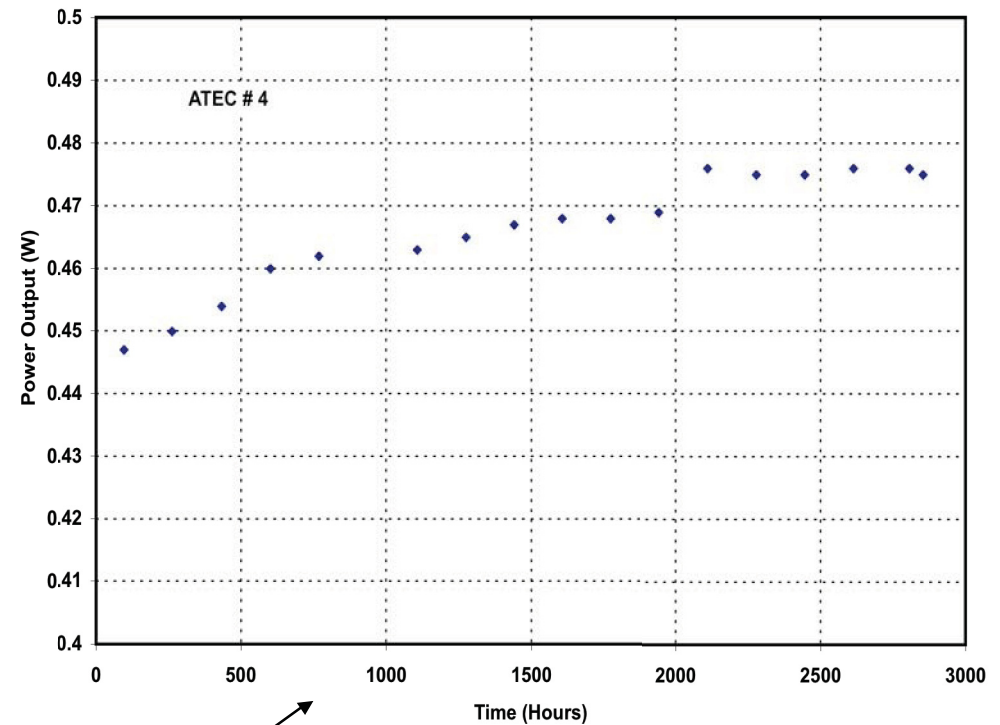
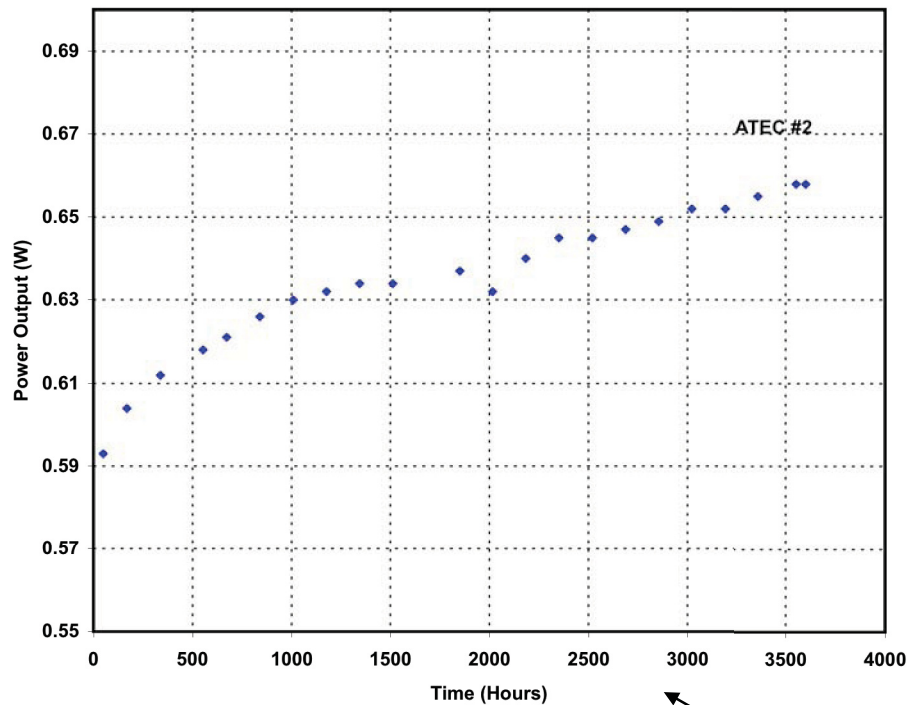


◆ Initiated performance life testing of spring-loaded low-T skutterudites uncouples

- In-gradient testing with $T_{\text{Hot}} \sim 875 \text{ K} \pm 10\text{K}$ and $T_{\text{Cold}} = 300 \text{ K} \pm 10\text{K}$
- Vacuum environment (10^{-6} Torr)
- Power output vs. load current at constant hot-shoe temperature



Low-T skutterudite unicouple performance life testing



■ ATEC 2:

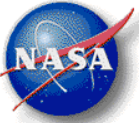
- ◆ P-leg: $\text{CeFe}_3\text{Ru}_1\text{Sb}_{12}$
- ◆ N-leg: CoSb_3
- ◆ $T_{\text{hot}} \sim 925\text{K}$
- ◆ $T_{\text{cold}} \sim 300\text{K}$



Skutterudite unicouple
with Nb hot-shoe

■ ATEC 4:

- ◆ P-leg: $\text{CeFe}_3\text{Ru}_1\text{Sb}_{12}$
- ◆ N-leg: CoSb_3
- ◆ $T_{\text{hot}} \sim 875\text{K}$
- ◆ $T_{\text{cold}} \sim 300\text{K}$



Summary

■ Summary

- ◆ RTG's have enabled surface and deep space missions since 1961
 - 26 flight missions without any RTG failures
 - Mission durations in excess of 25 years
- ◆ Future NASA missions require RTG's with high specific power and high efficiency, while retaining long life (> 14 years) and high reliability
 - 6-8 W/kg, 10-15% efficiency
- ◆ JPL in partnership with NASA-GRC, NASA-MSFC, DOE, Universities and Industry is developing advanced thermoelectric materials and converters to meet future NASA needs

■ Acknowledgements

- ◆ NASA Science Missions Directorate/ Radioisotope Nuclear Systems and Technologies